Amendments To the Abstract:

Please amend the abstract of the English translation document at page 12 lines 3 to 26, as follows:

Turbine, in particular a gas turbine, and a ~~moving~~ blade

The invention relates to a turbine ~~(1)~~ with a rotationally symmetrical rotor ~~(4)~~ on rotary bearings which has on its outer perimeter ~~moving~~ blade retaining grooves running transverse to the outer perimeter and set at the same angular distance from one another, having a ~~moving~~ blade ~~(13)~~ that has a ~~vane~~ wing ~~(24)~~ and a ~~moving~~ blade foot ~~(19)~~ which corresponds to the ~~moving~~ blade retaining groove ~~(18)~~ and which is inserted in one of the ~~moving~~ blade retaining grooves ~~(18)~~, wherein a platform ~~(23)~~ is located between the ~~moving~~ blade foot ~~(19)~~ and the ~~vane~~ wing ~~(24)~~ and extends transverse to the ~~moving~~ blade foot ~~(19)~~, having a baffle cooling element ~~(22)~~ which is located between two directly adjacent ~~moving~~ blades ~~(13)~~ between its platforms ~~(23)~~ and the rotor ~~(4)~~ and through which a cooling fluid can flow to cool the platform ~~(23)~~. It is proposed that the baffle cooling elements ~~(22)~~ are each formed from two independently moveable halves ~~(20, 21)~~ and that means are provided through which the two halves ~~(20, 21)~~ of the baffle cooling elements ~~(22)~~ are automatically pressed tight against each other by centrifugal force during turbine operation.

~~FIG 1~~